Masa - In- 83481

NASA Technical Memorandum 83481

NASA-TM-83481 19840003763

# Embedding Methods for the Steady Euler Equations

Shih-Hung Chang Cleveland State University Cleveland, Ohio

and

Gary M. Johnson

Lewis Research Center

Cleveland, Ohio

LIBRARY GOPY

. Nr. 23 1994

LIBRARY, NASA
HAMPTON, VIRGINIA

July 1983



		•
	:	

43 -1 RN/NASA-TM-83481 DISPLAY 43/2/1

84N11831\*\* ISSUE 2 PAGE 279, CATEGORY 64 RPT\*: NASA-TM-83481 E-1803 WAS 1.15:83481 CNT\*: NAG3-339 83/07/00 12 PAGES UNCLASSIFIED

DOCUMENT

UTTL: Embedding methods for the steady Euler equations

AUTH: AZCHANG, S. H.; BZJOHNSON, G. W.

CORP: Mational Aeronautics and Space Administration. Lewis Research Center,

Cleveland, Onio. AVAIL.NTIS SAP: HC A02/MF A01

MAJS: / \*EMBEDDING/ \*EULER EQUATIONS OF MOTION/ \*PROBLEM SOLVING

MINS: / CAUCHY-RIEMANN EQUATIONS/ COMPUTATIONAL FLUID DYNAMICS/ OPERATORS

(MATHEMATICS)

ABA: M. G.

ABS: An approach to the numerical solution of the steady Euler equations is to embed the first-order Euler system in a second-order system and then to

recapture the original solution by imposing additional boundary conditions. Initial development of this approach and computational experimentation with it were previously based on heuristic physical

reasoning. This has led to the construction of a relaxation procedure for the solution of two-dimensional steady flow problems. The theoretical

Justification for the embedding approach is addressed. It is proven that, with the appropriate choice of embedding operator and additional boundary conditions, the solution to the embedded system is exactly the one to the

original Euler equations. Hence, solving the embedded version of the Euler

EMTER:

	:	•		
			6 ·	
			. •	
	•			
		•		
		*		
				•
	•			
		* .	•	
		•		•
			•	
			•	
·			· · · · · · · · · · · · · · · · · · ·	
	-			

## EMBEDDING METHODS FOR THE STEADY EULER EQUATIONS

Shih-Hung Chang\*
Cleveland State University
Cleveland, Ohio 44115

and

Gary M. Johnson National Aeronautics and Space Administration Lewis Research Center Cleveland, Ohio 44135

#### **SUMMARY**

A recent approach to the numerical solution of the steady Euler equations is to embed the first-order Euler system in a second-order system and then to recapture the original solution by imposing additional boundary conditions. Initial development of this approach and computational experimentation with it have been based on heuristic physical reasoning. This has led to the construction of a relaxation procedure for the solution of two-dimensional steady flow problems. In the present report the theoretical justification for the embedding approach is addressed. It is proven that, with the appropriate choice of embedding operator and additional boundary conditions, the solution to the embedded system is exactly the one to the original Euler equations. Hence, solving the embedded version of the Euler equations will not produce extraneous solutions.

## INTRODUCTION

In the development of numerical solution procedures for the steady Euler equations, the common approach is to replace the steady equations by their unsteady counterparts and then to seek a temporally-asymptotic steady solution, either in real time ([1], [2]) or in pseudo time ([3] - [5]). Due to the difficulties associated with the numerical solution of a direct finite difference representation of the steady Euler equations, relatively few departures from this approach are to be found in the literature. Steger and Lomax [6] developed an iterative procedure for solving a nonconservation form of the steady Euler equations for subcritical flow with small shear. Desideri and Lomax [7] investigated preconditioning procedures on the matrix system arising from the finite differencing of the Euler equations. Bruneau, Chattot, Laminie and Guiu-Roux [8] have used a variational approach to transform the Euler equations into an equivalent second-order system. Preliminary numerical results have been presented for two-dimensional internal flows. Recently, Jespersen [9] has made significant progress toward adapting multigrid techniques to the solution of the Euler equations and has presented results for transonic flows over airfoils.

N84-1183#

<sup>\*</sup>Summer Faculty Fellow at Lewis Research Center, 1982-1983 (work partially funded by NASA Grant NAG3-339).

Johnson [10] - [12] proposed a surrogate-equation technique, in which the first order steady Euler equations are embedded in a second order system of equations. The solution of the original Euler equations is then recaptured by imposing appropriate additional boundary conditions on the embedded system. The advantages of such an approach are that the difficulties of solving the direct difference representation of the steady Euler equations can be bypassed and the resulting second-order embedded system can be solved by a variety of well-proven numerical procedures. Initial development of this approach and computational experimentation with it have been based on heuristic physical reasoning. This has led to the construction of a relaxation procedure for the solution of two-dimensional steady flow problems. All the numerical results in [10] - [12] suggest that this is a viable and potentially more economical approach than the alternative of solving the unsteady equations of motion.

In this report the theoretical justification for the embedding approach is addressed. It is proven that, with the appropriate choice of embedding operator and additional boundary conditions, the solution to the embedded system is exactly the one to the original Euler equations. Hence, solving the embedded version of the Euler equations will not produce extraneous solutions. The following section contains the proof for the two-dimensional Euler equations and shows that for the Cauchy-Riemann equations a similar result follows immediately. Generalizations to three dimensions and to other systems of partial differential equations are mentioned subsequently.

#### **EMBEDDING PROCEDURE**

The steady Euler equations can be written in vector form as

$$f_X + g_Y = 0 \tag{1}$$

where x and y are Cartesian coordinates,

$$f = \begin{bmatrix} \rho u \\ \rho u^2 + p \\ \rho uv \\ (E + p)u \end{bmatrix} \qquad \text{and} \qquad g = \begin{bmatrix} \rho v \\ \rho uv \\ \rho v^2 + p \\ (E + p)v \end{bmatrix}$$

Here  $\rho$ , p, u, v, and E denote respectively the density, static pressure, velocity components in the x and y directions, and the total energy per unit volume. Furthermore

$$E = \rho [e + 1/2 (u^2 + v^2)]$$

where the specific internal energy e is related to the pressure and density by the gas law

$$p = (\gamma - 1)\rho e$$

with  $\gamma$  denoting the ratio of specific heats.

Let
$$W = \begin{bmatrix} \rho \\ \rho u \\ \rho v \\ E \end{bmatrix}$$

By Euler's theorem on homogeneous functions, f and g can be expressed (see, for example, [13], [14]) as f = Aw and g = Bw, where A and B are the Jacobian matrices

$$A \equiv \frac{af}{aw}$$

and 
$$B \equiv \frac{\partial g}{\partial w}$$

We have

have 
$$A = -\begin{bmatrix} 0 & -1 & 0 & 0 \\ \frac{3-\gamma}{2} u^2 + \frac{1-\gamma}{2} v^2 & (\gamma - 3)u & (\gamma - 1)v & 1-\gamma \\ uv & -v & -u & 0 \\ \frac{\gamma E u}{\rho} + (1-\gamma)u(u^2 + v^2) & -\frac{\gamma E}{\rho} + \frac{\gamma - 1}{2}(3u^2 + v^2) & (\gamma - 1)uv & -\gamma u \end{bmatrix}$$

and

$$B = -\begin{bmatrix} 0 & 0 & -1 & 0 \\ uv & -v & -u & 0 \\ \frac{3-\gamma}{2}v^2 + \frac{1-\gamma}{2}u^2 & (\gamma-1)u & (\gamma-3)v & 1-\gamma \\ \frac{\gamma vE}{\rho} + (1-\gamma)v(u^2+v^2) & (\gamma-1)uv & -\frac{\gamma E}{\rho} + \frac{\gamma-1}{2}(3v^2+u^2) & -\gamma v \end{bmatrix}$$

Now, Eq. (1) can be written as

$$\frac{\partial}{\partial x}$$
 (Aw) +  $\frac{\partial}{\partial y}$  (Bw) = 0

or

$$\left[\frac{\partial}{\partial x} (A) + \frac{\partial}{\partial y} (B)\right] w = 0$$
 (2)

Let L denote the differential operator

$$L = \frac{\partial}{\partial x} (A) + \frac{\partial}{\partial y} (B)$$
 (3)

Then the Euler equations become

$$LW = 0 (4)$$

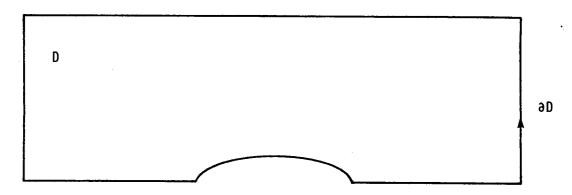
Now, let L\* be the formal adjoint operator to L defined by

$$L^* = -\left(A^T \frac{\partial}{\partial x} + B^T \frac{\partial}{\partial y}\right) \tag{5}$$

where  $\mathbf{A}^T$  and  $\mathbf{B}^T$  are the transposes fo A and B respectively. We may then consider the Euler equations (4) as embedded in the second-order system

$$L^* LW = 0 \tag{6}$$

Let D be a bounded closed region with a piecewise smooth boundary. For simplicity of argument, assume that Eq. (6) is defined in a domain containing D.



We now show that with an additional condition on the boundary, aD, of D, solutions of Eq. (6) are also solutions of Eq. (4).

 $\underline{\text{Theorem}}$  Let L and L\* be defined as in (3) and (5) respectively. If w is a solution of

$$L*Lw = 0$$
 in D

and satisfies the additional requirement

$$Lw = 0$$
 on  $\partial D$ ,

then it is also a solution of

$$Lw = 0$$
 in D.

<u>Proof</u> Let  $\langle \cdot, \cdot \rangle$  denote the Euclidean inner product in four-dimensional space. It can be shown that (see Appendix A for details) for any w,

$$\langle Lw, Lw \rangle - \langle w, L^*Lw \rangle = \frac{\partial}{\partial x} \langle Aw, Lw \rangle + \frac{\partial}{\partial y} \langle Bw, Lw \rangle$$
 (7)

Integrating over D and using Green's theorem, we obtain

$$\iint_{D} (\langle Lw, Lw \rangle - \langle w, L*Lw \rangle) \, dxdy$$

$$= \iint_{D} \left( \frac{\partial}{\partial x} \langle Aw, Lw \rangle + \frac{\partial}{\partial y} \langle Bw, Lw \rangle \right) \, dxdy$$

$$= \iint_{D} (\langle Aw, Lw \rangle dy - \langle Bw, Lw \rangle dx)$$
(8)

Here the line integral in the last expression of Eq. (8) is evaluated in the counterclockwise direction over the closed contour  $\partial D$ . Now, if w satisfies the hypotheses of the theorem, i.e. L\*Lw = 0 in D and Lw = 0 on  $\partial D$ , then Eq. (8) reduces to

$$\iint\limits_{\Omega} \langle Lw, Lw \rangle dxdy = 0$$

This implies that

$$\langle LW, LW \rangle = 0$$

in D and hence

$$Lw = 0$$

in D.

Q.E.D.

Now, consider the special case of the Cauchy-Riemann equations

$$u_X + v_V = 0 (9)$$

$$v_X - u_y = 0 \tag{10}$$

Let

$$f = \begin{bmatrix} u \\ v \end{bmatrix}, \qquad g = \begin{bmatrix} v \\ -u \end{bmatrix}$$

and rewrite Eqs. (9) and (10) in vector form

$$f_X + g_y = 0 \tag{11}$$

If we choose

then we have

$$A = \frac{af}{aw} = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}$$

and

$$B = \frac{\partial g}{\partial w} = \begin{bmatrix} 0 & 1 \\ -1 & 0 \end{bmatrix}$$

Eq. (11) can then be expressed as

$$\frac{\partial}{\partial x}$$
 (Aw) +  $\frac{\partial}{\partial y}$  (Bw) = 0

or

$$\left[\frac{\partial}{\partial x} (A) + \frac{\partial}{\partial y} (B)\right] w = 0$$

Hence, if we again use L to denote the differential operator

$$L = \frac{\partial}{\partial x} (A) + \frac{\partial}{\partial y} (B)$$
 (12)

the Cauchy-Riemann equations can also be written as

$$Lw = 0 (13)$$

Let

$$L^* = -\left(A^{\mathsf{T}} \frac{\partial}{\partial x} + B^{\mathsf{T}} \frac{\partial}{\partial y}\right) \tag{14}$$

Then Eq. (13) can be considered as embedded in

$$L^*Lw = 0 \tag{15}$$

Note that a few simple matrix multiplications will reduce Eq. (15) to

$$\frac{a^2}{ax^2} w + \frac{a^2}{ay^2} w = 0 {16}$$

which demonstrates simply the well-known fact that Eqs. (9) and (10) are embedded in the second-order system (16).

Now, let D be the same region as defined previously. Then the introduction of the differential operators L and L\* for the Cauchy-Riemann equations suggests the following immediate consequence of the previous theorem.

Corollary If w is a solution of Eq. (16) in D and if, on the boundary of D, it satisfies Eqs. (9) and (10), then it is also a solution of Eqs. (9) and (10) in D.

Thus if one wishes to obtain the unique solution to a boundary value problem of the Cauchy-Riemann equations (9) and (10), one can also solve Eq. (16) together with the original boundary conditions and the additional requirement that Eqs. (9) and (10) be satisfied on the boundary.\*

#### **GENERALIZATIONS**

Generalization of the result discussed above to the three-dimensional steady Euler equations is straightforward. Suppose the equations of motion are expressed as

$$f_X + g_V + h_Z = 0$$

This can then be written as

$$\left[\frac{\partial}{\partial x} (A) + \frac{\partial}{\partial y} (B) + \frac{\partial}{\partial z} (C)\right] q = 0$$

and hence the operator L can be introduced

$$L = \frac{\partial}{\partial x} (A) + \frac{\partial}{\partial y} (B) + \frac{\partial}{\partial z} (C)$$

The further details are analogous to the case of two dimensions.

The embedding concept can be used on any partial differential equations expressible in the form

$$Lq = 0$$
 or  $Lq = f$ 

However, we shall not pursue this idea further here.

In Desideri and Lomax [7], preconditioning matrices are investigated. In our Eq. (6), L\* may be considered as a preconditioning operator. Hence, the embedding method is a preconditioning procedure for the continuous model, while Desideri and Lomax's approach is one for the corresponding discrete model.

Based on the mathematical formulation presented here, two-dimensional steady Euler solvers are currently being developed. A detailed description of this work, including numerical results, will be presented in a forthcoming report.

### CONCLUSIONS

It has been proven that, for the numerical solution of the two-dimensional steady Euler equations, one can solve a second-order embedded system together

<sup>\*</sup>The authors understand that Dr. T. N. Phillips of ICASE - NASA Langley Research Center has recently obtained a similar result for the non-homogeneous Cauchy-Riemann equations.

with appropriate additional boundary conditions. This provides a theoretical justification for the recent computational experimentation with the surrogate-equation technique.

The proof presented here is extendible to three dimensions and the embedding technique is applicable to a wider class of partial differential equations than the Euler equations of motion considered here.

#### REFERENCES

- 1. Magnus, R.; and Yoshihara, H.: Inviscid Transonic Flow Over Airfoils. AIAA J., vol. 8, no. 12, Dec. 1970, pp. 2157-2162.
- 2. Warming, R.F.; and Beam, R.M.: Upwind Second-Order Difference Schemes and Applications in Aerodynamic Flows. AIAA J., vol. 14, no. 9, Sep. 1976, pp. 1241-1249.
- 3. Couston, M.; McDonald, P.W.; and Smolderen, J.J.: The Damping Surface Technique for Time-Dependent Solutions to Fluid Dynamic Problems. TN-109, von Karman Institute for Fluid Dynamics (Belgium), Mar. 1975.
- 4. Essers, J.A.: Methodes Nouvelles pour le Calcul Numerique d'Ecoulements Stationnaires de Fluides Parfaits Compressibles. Thèse de Doctorat en Sciences Appliquées, Université de Liege (Belgium), 1977.
- 5. Veuillot, J.P.; and Viviand, H.: Pseudo-Unsteady Method for the Computation of Transonic Potential Flows. AIAA J., vol. 17, no. 7, July 1979, pp. 691-692.
- 6. Steger, J.L.; and Lomax, H.: Calculation of Inviscid Shear Flow Using a Relaxation Method for the Euler Equations. Aerodynamic Analyses Requiring Advanced Computers, NASA SP-347, Pt. 2, Mar. 1975, pp. 811-838.
- 7. Desideri, J.A.; and Lomax, H.: A Preconditioning Procedure for the Finite-Difference Computation of Steady Flows. AIAA Paper 81-1006, June 1981.
- 8. Bruneau, C.H.; et al.: Finite Element Least Square Method for Solving Full Steady Euler Equations in a Plane Nozzle. Proceedings of the Eighth International Conference on Numerical Methods in Fluid Dynamics, Krause, E., ed., Lecture Notes in Physics, vol. 170, Springer-Verlag, 1982, pp. 161-166.
- 9. Jespersen, D.C.: A Multigrid Method for the Euler Equations. AIAA Paper 83-0124, Jan. 1983.
- 10. Johnson, G.M.: A Numerical Method for the Iterative Solution of Inviscid Flow Problems. Thèse de Doctorat en Sciences Appliquées, Université Libre de Bruxelles (Belgium), 1979.
- 11. Johnson, G.M.: Surrogate-Equation Technique for Simulation of Steady Inviscid Flow. NASA TP-1866, Sep. 1981.

- 12. Johnson, G.M.: Relaxation Solution of the Full Euler Equations. Proceedings of the Eighth International Conference on Numerical Methods in Fluid Dynamics, Krause, E., ed. Lecture Notes in Physics, vol. 170, Springer-Verlag, 1982, pp. 273-279.
- 13. Beam, R.M.; and Warming, R.F.: An Implicit Finite-Difference Algorithm for Hyperbolic Systems in Conservation-Law Form. J. Comput. Phys., vol. 22, no. 9, Sep. 1976, pp. 87-110.
- 14. James, G. and James, R.C., eds.: Mathematics Dictionary. Third ed. Van Nostrand-Reinhold, 1968.

# APPENDIX A

Derivation of Eq.(7) For any differentiable vector-valued functions U and V, we have

$$\left\langle \frac{\partial x}{\partial U}, V \right\rangle = -\left\langle U, \frac{\partial x}{\partial V} \right\rangle + \frac{\partial x}{\partial U} \left\langle U, V \right\rangle$$

and

$$\left\langle \frac{\partial U}{\partial y}, V \right\rangle = -\left\langle U, \frac{\partial V}{\partial y} \right\rangle + \frac{\partial}{\partial y} \left\langle U, V \right\rangle$$

Hence, we have

$$\langle Lw, Lw \rangle = \langle \frac{\partial}{\partial x} (Aw) + \frac{\partial}{\partial y} (Bw), Lw \rangle$$

$$= \langle \frac{\partial}{\partial x} (Aw), Lw \rangle + \langle \frac{\partial}{\partial y} (Bw), Lw \rangle$$

$$= -\langle Aw, \frac{\partial}{\partial x} (Lw) \rangle + \frac{\partial}{\partial x} \langle Aw, Lw \rangle - \langle Bw, \frac{\partial}{\partial y} (Lw) \rangle + \frac{\partial}{\partial y} \langle Bw, Lw \rangle$$

$$= -\langle W, A^{T} \frac{\partial}{\partial x} (Lw) \rangle + \frac{\partial}{\partial x} \langle Aw, Lw \rangle - \langle W, B^{T} \frac{\partial}{\partial y} (Lw) \rangle + \frac{\partial}{\partial y} \langle Bw, Lw \rangle$$

$$= \langle W, -\langle A^{T} \frac{\partial}{\partial x} + B^{T} \frac{\partial}{\partial y} \rangle LW \rangle + \frac{\partial}{\partial x} \langle Aw, Lw \rangle + \frac{\partial}{\partial y} \langle Bw, Lw \rangle$$

$$= \langle W, L^{*}Lw \rangle + \frac{\partial}{\partial x} \langle Aw, Lw \rangle + \frac{\partial}{\partial y} \langle Bw, Lw \rangle$$

This is Eq. (7).

•			
,			
£			
-			

.

4.5			<u> </u>		
1. Report No. NASA TM-8348]	2. Government Accession	on No.	3. Recipient's Catalog N	0.	
4. Title and Subtitle			5. Report Date		
Embedding Methods for the	Stoody Fulor For		July 1983		
Time adding free chods for the	lations	6. Performing Organizati	on Code		
	•		505-31-0		
7. Author(s)		1	3. Performing Organizati	on Report No.	
Shih-Hung Chang and Gary M. Johnson			F 1002		
Sittle liding chang and Gary M. Johnson		10	E-1803 D. Work Unit No.		
9. Portornico Organization Name and Address					
9. Performing Organization Name and Address National Aeronautics and S	nace Administmat	tion 11	. Contract or Grant No.		
Lewis Research Center	pace Auministrat	, ion	·		
Cleveland, Ohio 44135			3. Type of Report and Pe	riod Covered	
12. Sponsoring Agency Name and Address			T   1   1   1   1   1   1   1   1   1	_	
National Aeronautics and S	pace Administrat	ion 14	Technical Memorandum  14. Sponsoring Agency Code		
Washington, D.C. 20546			. oponsoring Agency Ot		
15. Supplementary Notes			<del></del>		
Shih-Hung Chang, Cleveland Fellow at Lewis Research Co NAG3-339); Gary M. Johnson	enter. 1982-1983	(Work nartially	nio and Summer V funded by NA	Faculty SA Grant	
A recent approach to the numbed the first-order Euler the original solution by in ment of this approach and con heuristic physical reason procedure for the solution report the theoretical just is proven that, with the appoundary conditions, the so original Euler equations. tions will not produce extra	r system in a senposing addition computational exponing. This has of two-dimension for topropriate choiculution to the enders.	cond-order systemal boundary concept of the constant of the constant of the constant of the constant of the embedding of the embedded years.	em and then to ditions. Init the it have be truction of a problems. In proach is address avactly the	recapture ial develop- en based relaxation the present essed. It dditional	
17. Key Words (Suggested by Author(s))		18. Distribution Statement			
Numerical analysis; Compute	rized	Unclassified -			
simulations; Steady flow; C flow; Euler equations of mo	ompressible	STAR Category			
19. Security Classif. (of this report)	20. Security Classif. (of this	l page)	21. No. of pages	22. Price*	
Unclassified	Unclassifie	ł			

*		
•		
r		

National Aeronautics and Space Administration

Washington, D.C. 20546

Official Business
Penalty for Private Use, \$300

SPECIAL FOURTH CLASS MAIL BOOK





Postage and Fees Paid National Aeronautics and Space Administration NASA-451

NASA

POSTMASTER:

If Undeliverable (Section 158 Postal Manual) Do Not Return